

RAC High Precision Satellite Positioning Receiver Specification RAC high precision GNSS receiver

Achieve high-precision satellite positioning without differential stations and ground-based augmentation networks

Real-time Array Calibration

Product Model:RAC-C1

Hardware version:RAC-D0030124R_V1.0

Software Version:GLN2-BSS-UV1.23-180602



Document Revision History

Version	Revision Date	Version described
V1.0	2016/5/18	New Document
V1.1	2016/12/15	Performance index parameters increase
V1.2	2017/6/18	Enter instructions to optimize changes
V2.0	2021/6/09	Detail optimization

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1. Product Description

1.1Product Overview

Product Name:RAC-C1 high-precision satellite positioning receiver

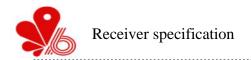


Figure 1.1 Appearance of RAC-C1 high-precision satellite positioning receiver

This product is a G-MOUSE-like satellite receiver (hereinafter referred to as G-MOUSE), which is a complete GPS satellite positioning receiver; it has a built-in satellite receiving antenna and has a full range of high-precision positioning functions. The positioning accuracy can reach less than 1.5 meters in static and less than 0.6 meters in dynamic; it can meet the strict requirements of industrial-grade positioning and personal use needs. Through technological innovation, the product has two core competitive advantages, namely, no differential station is required, no L2 or B3 precision code is used, and the cost is low . Unlike traditional high-precision satellite positioning technology, our technology gets rid of the differential technology that high-precision satellite positioning relies on, which greatly reduces the cost. This is our most important advantage over similar products in the world. This product has the characteristics of high precision, high sensitivity, low power consumption, and miniaturization. Its extremely high tracking sensitivity greatly expands its positioning coverage.

1.2 Product Features

- ◆ UART (3.3V_TTL level)/RS232/IIC/CAN interface output, faster application;
- ◆ Using KDS 0.5PPM high-precision TCXO;
- ◆ Rich data output rate: 115200bps (default) [optional: 9600/38400];
- ◆ Output statement: NMEA 0183 V3.0/UBX protocol;
- ◆ Support adjustable data refresh rate: 1Hz-10Hz;
- ◆ Independently develop and design the antenna vibrator to ensure that the phase center coincides with the geometric center, minimizing the impact of the antenna on measurement errors;
- ◆ Support optional A-GPS service, inertial navigation, geomagnetic sensor, and pressure sensor;
- ◆ GPS, BD, GLONASS hybrid engine optional;
- ◆ Manufactured using lead-free technology, compliant with RoHS standards.



1.3 Certification Testing

- ◆ The positioning chip has passed AEC-Q100 certification.
- ◆ The positioning module has passed ISO16750 test.
- ◆ Positioning products have passed ISO7637 test.
- ◆ The production line complies with ISO/TS-16949 certification.

1.4 Application

- ◆ Unmanned driving, positioning navigation and tracking products;
- ◆ UAV field, agricultural plant protection machine, precision agriculture;
- ◆ Synchronize UTC time and timing field;
- ◆ Widely used in waterway surveying and ocean surveying;
- ◆ Products such as track recording and GPS data point calibration;
- ◆ Surveying and mapping products such as high-precision mapping, area measurement and distance measurement;
- ◆ Road construction, earthquake monitoring, bridge deformation monitoring, landslide monitoring, terminal container operations, etc.

1.5 Performance Indicators

Sensitivity	Track	- 165dBm	
Schsitivity	Capture	- 148dBm	
	Cold Start	Average 42S (Open Sky)	
TTFF (Time to First Fix)	Warm start	Average<5s	
(11110 to 11101111)	Hot Start	Average<1s	
	22 Tracking chan	nels, 56 Capture channels	
Туре	Frequency 42MHzBI	1 1575. OS B1 1561. C/Acode MHz	
Positioning accuracy	<1.5m	Static	
	<0.6m	Dynamic	
	Coordinate system	Coordinate basis:WGS-84	

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Speed accuracy 0.1 m/sTime pulse 1s (default) (configurable) Update rate 1-10Hz (Optional) 100ms (1pps=1Hz=1times/second) **PPS** High <18000m Operational restrictions Speed <100m/s Acceleration <8g NMEA 0183 The default output information of the general protocol is: NMEA-0183-Data Format RMC

Power Specifications

Power supply	DC 5.0V (Ripple voltageVpp<200mv)
Working current	Track:160mA Capture:180mA Peak:200mA
Standby current	<5uA
Functional Interface	UART/IIC/RS232/CAN

Physical properties

Temperature	Operating temperature	- 30°C~80°C	
	Storage temperature	- 40°C~85°C	

1.6 Product port definition

The product output can support UART or IIC output, etc. The default is UART output (IIC, etc. can be expanded).

The specific port definition is shown in the figure below:

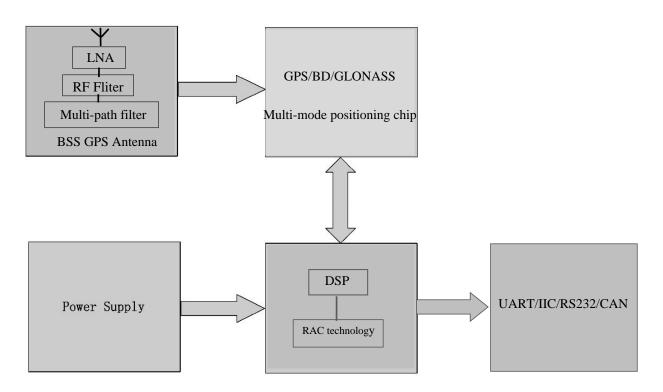
Pin NO.	Pin Name	I/O	Remark
1	GND	land	Ground (yellow line)
2	VCC	power supply	DC 5.0V Peak current 206mA (Red line)
3	RX	enter	TTL Serial port receiving (blue line)
4	TX	output	TTL Serial port send (white line)
5	SDA	data	Reserved geomagnetic IIC data (green)



6 SCL clock	Reserved geomagnetic IIC clock (orange)
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surfae5.1 IICReserved port definition

1.7 Functional Block Diagram



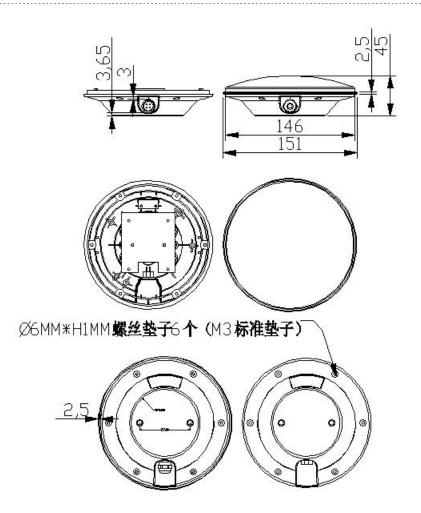
The RAC-C1 high-precision positioning receiver uses a high-precision antenna and independently developed antenna oscill -ators to ensure that the phase center coincides with the geometric center, minimizing the impact of the antenna on measurement err -ors. The antenna receives satellite signals in the air and transmits them to the chip RF unit, which tansmits the signals to the chip ba -seband unit. After a series of analysis, it is transmitted to the DSP. After being processed by RAC's independently developed algori -thm, high-precision positioning data and 1PPS signals are output through the serial port, RS232, and CAN interfaces, and the IIC outputs geomagnetic information.

1.8 Product size

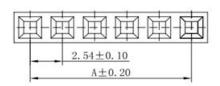
Overall size: Length151mm*high45mm (There is a strong magnet inside the bottom shell)

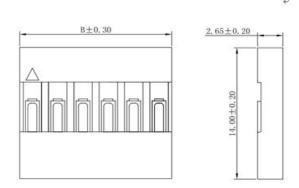
PCBSize: Diameter120mm*high12mm

Bottom case screw specifications: Standard M5 Specifications (full thread)



1.9 Interface terminal size





H2544-04: 7.62*10.16mm

1*4P spacing 2.54mm DuPont Terminal



2. Input and Output Statements

2.1 Input Statement

2.1.1 Instruction Format

BSS-(command 1)-(command 2)-(command 2)....(command n) (CR) (LF)

2.1.2 Instruction Description

- ◆ BSS is the instruction header, and each instruction must start with BSS:
- ◆ Each command must be separated by a "-" (hyphen, i.e., minus sign);
- ◆ Each instruction needs to be ended with a "carriage return line feed";
- ◆ When NMEA0183 protocol mode is selected, GPRMC outputs by default;
- ◆ In UBX protocol, 5 pieces of information including posllh, sol, dop, velned and status are output by default;
- ◆ To switch from NMEA protocol to UBX protocol or from UBX protocol to NMEA protocol, you need to power on again. If there is a battery to save the ephemeris, you need to turn off the power for several hours and wait until the battery is exhausted before it works.

2.1.3 Command List

Order	Illustrate
0183	Choose one of the two, indicating that the NMEA0183 protocol is selected
UBX	Choose one of the two, indicating that the UBX protocol is selected
115200	Select one of the three options, indicating the serial port baud rate is 115200
38400	Select one of the three options, indicating the serial port baud rate is 38400
9600	Select one of the three options, indicating the serial port baud rate is 9600
10HZ	Select one of the three options, indicating that the serial port output frequency is 10HZ
5HZ	Select one of the three options, indicating that the serial port output frequency is 5HZ
1HZ	Select one of the three options, indicating that the serial port output frequency is 1HZ
RHXZ	Indicates that the geomagnetic sensor is enabled.
D10	Choose one of the three, indicating that the number of geomagnetic outputs in one frame of GPS time is 10 times
D5	Choose one of the three, indicating that the number of geomagnetic outputs in one frame of GPS time is 5 times



D1	Choose one of the three, indicating that the number of geomagnetic outputs in one frame of GPS time is 1
VTG	GPVTG command, only enabled when NMEA0183 protocol is selected
GGA	GPGGA command, only enabled when NMEA0183 protocol is selected
PVT	PVT command, only enabled when UBX protocol is selected
	Only valid for NMEA0183 protocol, all commonly used GPS commands
ALL	are enabled, including RMC, VTG, GGA, GSA, GSV, to facilitate
	observation of satellite information

Note: Due to the relatively low data transmission rate of the serial port, the frequency of GPS data output and magnetometer output is closely related. Incorrect settings may result in data output errors or loss.

Simplified Commands: When outputting only the RMC command:

When the serial port baud rate is 9600, the serial output frequency is 5Hz, and the magnetometer output is D1.

When the serial port baud rate is 38400, the serial output frequency is up to 10Hz, and the magnetometer output is up to D10.

When the serial port baud rate is 115200, the serial output frequency is up to 10Hz, and the magnetometer output is up to D10.

Multiple Commands: When outputting multiple GPS commands:

When the serial port baud rate is 9600, the serial output frequency is 1Hz, and the magnetometer output is D1.

When the serial port baud rate is 38400, the serial output frequency is up to 5Hz, and the magnetometer output is up to D5.

When the serial port baud rate is 115200, the serial output frequency is up to 10Hz, and the magnetometer output is up to D10.

Examples:

Change to NMEA0183 protocol, baud rate 115200, frequency 10Hz, enable VTG, GGA, and magnetometer, magnetometer output as one frame, GPS outputs one magnetometer information:

BSS-0183-115200-VTG-GGA-10HZ-RHXZ-D1\r\n

Change to UBX protocol, baud rate 115200, frequency 10Hz, enable PVT and magnetometer, magnetometer output as one frame, GPS outputs one magnetometer information:

BSS-UBX-115200-10HZ-PVT-RHXZ-D1\r\n

After the modification is successful, the serial terminal will display as follows:

Seting parameter success Saving parameter Saving parameter success reseting system Please wait



2.2 Output Statement

2.2.1 NMEA 0183 protocol

NMEA 0183 is a standard format developed by the National Marine Electronics Association for marine electronic equipment. It has become a unified RTCM (Radio Technical Commission for Maritime services) standard protocol for GPS navigation equipment. The NMEA-0183 protocol uses ASCII code to transmit GPS positioning information. The data format protocol framework is as follows:

	位 5位	和的计算范围		
\$	<地址>	{,<数值>}	*<校验和>	<cr><lf></lf></cr>
起始符	地址段	数据段	校验和段	结束序列
每条语句 都是以 '\$' 开始	分为两部分: 发送器标识符 和语句类型	以','开始,后面的数值长度是可变的,也有是定长的	对' \$'和 ' *'之间的数 据(不包括这 两个字符)按 字节进行异或 运算的结果, 用十六进制数 值表示	每条语句都 是以 <cr><cf> 结束</cf></cr>

Statement output:

	Information	Illustrate
1	RMC	Recommended minimum dedicated navigation data
2	VTG	Ground speed and heading
3	GGA	Receiver positioning data
4	GSA	DOP and effective satellites
5	GSV	Visible satellites
6	GLL	Geographic location - Latitude/Longitude
7	RHXZ	Geomagnetic information

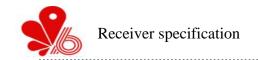
2.2.1.1 Minimum dedicated navigation data recommended by RMC

Information	RMC
Describe	Recommended minimum positioning information
Type	Output
Format	\$ RMC,UTCtime,status,lat,uLat,lon,uLon,spd,cog,UTCdate,mv,mvE,mode*CS <cr><lf></lf></cr>
Example	\$GNRMC,084103.00,A,2233.39544,N,11356.55665,E,0.035,,220618,,,A*7A

Parameter Description Parameter Description Fields name Format \$--RMC String Message ID, RMC statement header, '--' is the system identifier 1 UTC time hhmmss.ss The current location's UTC time 2 character Position valid flag. V = invalid data A = valid data3 status Latitude, the first two characters represent degrees, the following characters ddmm.mmmmm 4 lat represent minutes 5 Latitude direction: N-North, S-South uLat character Longitude, the first 3 characters represent degrees, the following characters dddmm.mmmmm 6 lon represent minutes uLon character Longitude direction: E-East, W-West 7 spd Ground speed, in knots Numeric 8 True heading over ground, in degrees cog Numeric 10 UTC date ddmmyy Date (dd for day, mm for month, yy for year) Numeric Magnetic declination, in degrees. Fixed to be empty mv 11 Magnetic declination direction: E-East, W-West. Fixed to empty character 12 mvE Mode indication (A = autonomous positioning, D = differential, R = RTK, E = 13 Mode character estimation, N = invalid data) Navigation status indicator (V means the device does not provide navigation 14 navStatus character status information)NMEA v4.1 and above only Checksum, the XOR result of all characters between \$ and * (excluding \$ and Hexadecimal value 15 CS Carriage return and line feed <CR><LF>character 16

2.2.1.2 VTG Ground Speed and Course

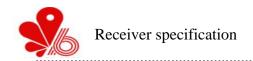
	<u>.</u>						
Inform	ation	VTG					
Desc	ribe	Speed over ground and heading over ground information.					
Тур	pe		Output				
Forr	nat	\$VTG,cogt,T,	.cogm,M,sog,N,kph,K,mode*CS <cr><lf></lf></cr>				
Exan	nple	\$GNVT	G,75.20,T,,M,0.009,N,0.017,K,A*02				
	<u>,</u>	Parame	eter Description				
Fields	name	Format	Parameter Description				
1	\$VTG	String	Message ID, VTG statement header, '' is the system identifier				
2	cogt	Numeric	Course relative to true north, in degrees				
3	Т	character	True north indication, fixed at T				
4	cogm	Numeric	Heading relative to magnetic north, in degrees				
5	M	character	Magnetic north indication, fixed at M				
6	sog	Numeric	Ground speed, in knots				
7	N	character	Speed unit is knot, fixed as N				
8	kph	Numeric	Ground speed in kilometers per hour				
9	K	character	Speed unit, kilometers per hour, fixed as K				
10	,		Mode indication (A = autonomous positioning, D = differential				
10	mode	character	, $R = RTK$, $E = estimation$, $N = invalid data$)				



11	CS	Hexadecimal value	Checksum, all characters between \$ and * (excluding \$ and *)XOR result
12	<cr><lf></lf></cr>	character	Carriage return and line feed

2.2.1.3 GGA receiver positioning data

Inform	ation			GGA			
Descr			Receiver time	, location and positioning related data			
Тур	e			Output			
Forn		GGA,UTCti	\$ ime,lat,uLat,lon,uLon,FS,numSv,HDOP,msl,uMsl,sep,uSep,diff Age,diffSta *CS <cr><lf></lf></cr>				
Exam	ple	\$GNGGA	,235316.00,2959.99	250,S,12000.00900,E,1,06,1.21,62.77,M, 0.00,M,,*7B			
			Parame	ter Description			
Fields		name	Format	Parameter Description			
1		\$GGA	String	Message ID, GGA statement header, '' is the system identifier			
2	Ţ	JTC time	hhmmss.ss	The current location's UTC time			
3		lat	ddmm.mmmmm	Latitude, the first two characters represent degrees, the following characters represent minutes			
4		uLat	character	Latitude direction: N-North, S-South			
5		lon dddmm.mmmmi		Longitude, the first 3 characters represent degrees, the following characters represent minutes			
6		uLon	character	Longitude direction: E-East, W-West			
7	FS		Numeric	GPS positioning status ($0 = \text{no positioning}$, $1 = \text{non-differential positioning}$, $2 = \text{differential positioning}$)			
8				The number of satellites used for positioning, 00~12			
9		HDOP	Numeric	Horizontal Dilution of Precision (HDOP<1.2 is high- precision positioning)			
10		msl	Numeric	Altitude, that is, the height of the receiver antenna relative to the geoid high			
11		uMsl	character	Altitude unit, meter, fixed character M			
12		sep	Numeric	The distance between the reference ellipsoid and the geoid, "-" indicates The geoid is lower than the reference ellipsoid			
13		uSep	character	Altitude unit, meter, fixed character M			
14	diffAge		Numeric	Differential time (from the last differential signal received)Seconds, non-differential positioning, this item is empty)			
15		diffStat	Numeric	The ID of the differential reference station (empty for non-differential positioning)			
16	CS		Hexadecimal value	Checksum, all characters between \$ and * (excluding \$ and *)XOR result			
17	<cr><lf></lf></cr>		character	Carriage return and line feed			



2.2.1.4 GSA DOP and effective satellites

on	GSA					
e	Satellite number and DOP information used for positioning.					
		Output				
\$GS	A,smode,FS{,SVID	},PDOP,HDOP,VDOP,systemId*CS <cr><lf></lf></cr>				
e	\$GNGSA,A,3,0	5,21,31,12,18,29,,,,,2.56,1.21,2.25,1*01				
	Param	eter Description				
name	Format	Parameter Description				
\$GSA	String	Message ID, GSA statement header, '' is the system identifier				
smode	character	Mode switch indication ($M = Manual, A = Automatic$)				
		Positioning status flag (1 = not positioned, $2 = 2D$				
FS	number	positioning, $3 = 3D$ positioning)				
{,SVID}	Numeric	Satellite number used for positioning				
PDOP	Numeric	Position Dilution of Precision (PDOP)				
HDOP	Numeric	Horizontal Dilution of Precision (HDOP)				
VDOP	Numeric	Vertical Dilution of Precision (VDOP)				
systemId	Numeric	GNSS system ID defined by NMEA (Note [1])NMEA v4.1 and above only				
CS	Hexadecimal value	Checksum, all characters between \$ and * (excluding \$ and *)XOR result				
<cr><lf></lf></cr>	character	Carriage return and line feed				
	Remark[1] GNSSsystemID				
ystem ID		Describe				
1		GPS system				
2		GLONASS system				
4	BDS system					
	name \$GSA smode FS {,SVID} PDOP HDOP VDOP systemId CS <cr><lf> vstem ID 1 2</lf></cr>	satellite number a \$GSA,smode,FS{,SVID} \$ \$GNGSA,A,3,0; Param name Format \$GSA String smode character FS number {,SVID} Numeric PDOP Numeric HDOP Numeric VDOP Numeric systemId Numeric CS Hexadecimal value <cr> <cr> <lf> character Remark[ystem ID 1 2</lf></cr></cr>				

2.2.1.5 GSV visible satellite information

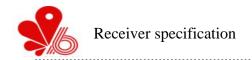
Inforn	nation	GSV				
Describe The satellite number and elevation, azimuth, carrier-to-noise ratio, etc. of the visatellite. The number of {satellite number, elevation, azimuth, carrier-to-noise ratio} parameter groups in each GSV sentence is varied with a maximum of 4 groups and a minimum of 0 groups.						
Ty	pe			Output		
For	mat	\$GSV,numMsg,msgNo,numSv{,SVID,ele,az,cn0}*CS <cr><lf></lf></cr>				
Exar	mple	\$GPGSV,3,1,12,02,39,117,25,04,02,127,,05,40,036,24,08,10,052,*7E \$GPGSV,3,2,12 09,35,133,,10,01,073,,15,72,240,22,18,05,274,*7B \$GPGSV,3,3,12,21,10,316,,24,16,176,,26,65,035,42,29,46,277,18*7A				
			Paramet	ter Description		
Fields		name	Format	Parameter Description		
1		\$GSV	String	Message ID, GSV statement header, '' is the system identifier		



Each GSV sentence outputs information about up to 4 visible satellites. Therefore, when the system has more 2 numMsg number than 4 visible satellites, multiple GSV statements will be required. msgNo number Current statement number 3 numSv Numeric Total number of visible satellites 4 They are: Satellite number; Elevation angle, ranging from 0 to 90, in degrees; Azimuth angle, ranging from 0 to 359, in degrees; {,SVID,ele, 5 Numeric az,cn0} Carrier-to-noise ratio, ranging from 0 to 99, in degrees. dB, if the current satellite is not tracked, the output is empty. GNSS signal ID defined by NMEA (0 represents all signalId 6 Numeric signals)NMEA v4.1 and above only Checksum, all characters between \$ and * (excluding Hexadecimal 7 CS \$ and *)XOR result value 8 <CR><LF>character Carriage return and line feed

2.2.1.6 GLL Geographic Location - Latitude/Longitude

Information			GLL				
D	Describe Informa		ation such as latitude, longitude, positioning time and positioning status.				
	Туре			Output			
F	Format		\$GLL,lat,uLat,lon,	uLon, UTCtime,valid,mode*CS <cr><lf></lf></cr>			
E	xample		\$GNGLL,2959.99	250,S,12000.00900,E,235316.00,A,A*4E			
			Parameter l	Description			
Fields	na	me	Format	Parameter Description			
1	\$0	GLL	String	Message ID, GLL statement header, '' is the system identifier			
2	lat		ddmm.mmmmm	Latitude, the first two characters represent degrees, the following characters represent minutes			
3	ul	Lat	character	Latitude direction: N-North, S-South			
4	lon		dddmm.mmmmm	Longitude, the first 3 characters represent degrees, the following characters represent minutes			
5	uI	Lon	character	Longitude direction: E-East, W-West			
6	UTC	time	hhmmss.ss	The current location's UTC time			
7	va	llid	character	Data status (A = valid positioning, V = invalid positioning)			
8	mode		character	Mode indication (A = autonomous positioning, D = differential, E = estimated, N = invalid data)			
9	CS		Hexadecimal value	Checksum, all characters between \$ and * (excluding \$ and *)XOR result			
10	<cr:< td=""><td>><lf></lf></td><td>character</td><td>Carriage return and line feed</td></cr:<>	> <lf></lf>	character	Carriage return and line feed			



2.2.1.7 RHXZ geomagnetic information

	Information		RHXZ					
	Describe		Geomagnetic sensor information					
	Type			Output				
	Format		\$R	HXZ,X,Y,Z*CS <cr><lf></lf></cr>				
	Example		\$I	RHXZ,0057,FE6E,0210*45				
			Parameter De	scription				
Fields	name		Format	Parameter Description				
1	\$RHX	Z	String	Message ID, RHXZ statement header				
2	X	X		The hexadecimal value of the geomagnetic sensor X axis (high bit firlsitk, e 0057 means 0x0057, range 0000 ~FFFF, the leading 0 will also be transmitted)				
3	Y		Hexadecimal value	The hexadecimal value of the Y axis of the geomagnetic sensor (highlike digit first, FE6E means 0 xFE6E, range 0000~FFFF, the leading 0 will also be transmitted)				
4	Z		Hexadecimal value	The hexadecimal value of the Z axis of the geomagnet sensor (highlike bit first, 0210 means 0x0210, range 0 00~FFFF, the leading 0 will also be transmitted)				
5	5 CS		Hexadecimal value	Checksum, all characters between \$ and * (excluding \$ and *)XOR result				
6	<cr><l< td=""><td>F></td><td>character</td><td>Carriage return and line feed</td></l<></cr>	F>	character	Carriage return and line feed				

2.2.2 UBXprotocol

The UBX protocol is a proprietary protocol developed by u-blox, and it has the following key characteristics:

Compact: Uses 8-bit binary data.

Checksum protection: Employs a low-overhead checksum algorithm. Modular: Uses a two-stage message identifier (Class and Message ID).

The structure of a UBX protocol packet is shown in Figure 1. As seen in the diagram, each message consists of three parts: header, data, and checksum. The header is two bytes: 0xB5 0x62, which identifies the start of a UBX protocol data transmission.

CLASS is one byte, representing the category of the test data message.

ID is one byte, indicating the specific parameter output under a given CLASS.

LENGTH indicates the length of the data section (in bytes).

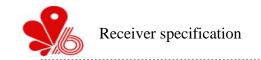
CK_A and CK_B are two checksum bytes, used for data verification from the CLASS to the Payload section

SYNC Cha r1 SYNCCh ar2	CLASS	ID	LENGTH (little endian)	PayLoad	CK_A	CK_B
------------------------------	-------	----	------------------------------	---------	------	------

The following introduces the commonly used UBX data packets.

1. UBX-NAV-POSLLH (0x01 0x02)

Message	NAV-POSLLH
---------	------------



Des	cription	Geodetic Position Solution						
Message Structure		Header	Class	ID	Length (Bytes) Payload Chec		Checksum	
		0xB5 0x62	0x01	0x02	28	see below	CK_A CK_ B	
			Paylo	oad Content	ts:			
Byte Off set	Number Format	Scaling	Name	Unit	Description			
0	U4	-	iTOW	ms	GPS time of week of the navigation epoch.			
4	I 4	1e-7	lon	Deg	Longitude			
8	I 4	1e-7	lon	Deg		Latitude		
12	I 4	-	height	mm	Height above ellipsoid			
16	I4	-	oeLh	mm	Height above mean sea level			
20	U4	-	Нс	mm	Horizontal accuracy estimate			
24	U4	-	vv	mm	Vertical accuracy estimate			

2. UBX-NAV-STATUS (0x01 0x03)

Message		NAV-STATUS							
De	escription		Receiver Navigation Status						
		Header	Class	ID	Length (Byt es)	Payload	Checksu m		
Messa	Message Structure		0x01	0x03	16	see below	CK_ACK_ B		
			Payload	Contents:					
Byte Offs et	Number For mat	Scaling	Name	Unit	Description				
0	U4	-	iTOW	ms	GPS time of week of the navigation epoch.				



GPSfix Type, this value does not qualify a fix as valid and within the limits. See note on flag gpsFixOk below. 0x00 = no fix4 U1 0x01 = dead reckoning onlygpsFix 0x02 = 2D-fix 0x03 = 3D-fix 0x04=GPS+dead reckoning combined 0x05 = Time only fix0x06..0xff = reservedflags 5 X1 Navigation Status Flags 6 X1 fixStat Fix Status Information further information about navigation 7 X1 flags2 output Time to first fix (millisecond time tag) 8 U4 ttff ms U4 12 Milliseconds since Startup/Reset msss mm

3. UBX-NAV-DOP (0x01 0x04)

M	essage		NAV-DOP							
Des	scription	Dilution of precision								
		Header	Class	ID	Length (Bytes)	Payload	Checksum			
Messag	Message Structure		0xB5 0x62 0x01 0x04		18	see below	CK_A CK_ B			
			Paylo	oad Conten	ts:					
Byte Off set	Number	Scaling	Name	Name Unit Description						



Format GPS time of week of the navigation 0 U4 iTOW ms epoch. 4 U2 0.01 gDOP Geometric DOP 6 U2 0.01 pDOP Position DOP -8 0.01 TDOP U2 Time DOP 10 U2 0.01 vDOP Vertical DOP 12 U2 0.01 Horizontal DOP HkDOP U2 0.01 nDOP 14 Northing DOP -0.01 eDOP 16 U2 **Easting DOP**

4. UBX-NAV-SOL (0x01 0x06)

N	Message						1	NAV-SOL			
De	escription				N	Javig	ation	Solution Info	ormation		
			Header Cl		ass		ID	Length (Bytes)	Payload	Checksum	
Messa	Message Structure		0x62 0x0		01 0x0		.06	52	see belo w	CK_A CK_B	
			Payload Contents:								
Byte O	Number	Scaling	Na	Name Unit		nit			Description		
ffset	Format	Scaling	144		O1	11t	Description				
0	U4	-	iTO)W	1	ms	GPS time of week of the navigation epoch.				
							Fractional part of iTOW (range: +/-500000).				
4	I 4	-		f		ns	The precise GPS time of week in seconds is:				
							(iTOW * 1e-3) + (fTOW * 1e-9)				
8	I2	-	we	week weeks		GPS week number of the navigation epoch					
10	U1	-	gps	Fix		-		GPSfix Type	e, range 05 (0x00 = No Fix	

					0x01 = Dead Reckoning only
					0x02 = 2D-Fix
					0x03 = 3D-Fix
					0x04 = GPS + dead reckoning combined
					0x05 = Time only fix
					0x060xff: reserved
11	X1	-	flags	1	Fix Status Flags
12	I4	-	eX	cm	ECEF X coordinate
16	I4	-	eCE	cm	ECEF Y coordinate
20	I4	-	eCEf	cm	ECEF Z coordinate
24	U4	-	pAcc	cm	3D Position Accuracy Estimate
28	I4	-	xvX	cm/s	ECEF X velocity
32	I4	-	VxD	cm/s	ECEF Y velocity
36	I4	-	VxD	cm/s	ECEF Z velocity
40	U4	-	sAcc	cm/s	Speed Accuracy Estimate
44	U2	0.01	pDOP	-	Position DOP
46	U1	-	Reserve d1	-	Reserved
47	U1	-	numSV	-	Number of SVs used in Nav Solution

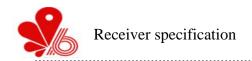
5. UBX-NAV-PVT (0x01 0x07)

Message	NAV-PVT								
Description		Navigation Position Velocity Time Solution							
	Header	Header Class ID Length (Bytes)		Payload	Checksum				
Message Structure	0xB5 0x62	0x01	0x07	92	see below	CK_A CK_ B			
Payload Contents:									



.....

Byte Offset	Number Format	Scalin g	Name	Unit	Description
0	U4	-	iTOW	ms	GPS time of week of the navigation epoch.
4	U2	-	year	у	Year (UTC)
6	U1	-	month	month	Month, range 112 (UTC)
7	U1	-	day	d	Day of month, range 131 (UTC)
8	U1	-	hour	h	Hour of day, range 023 (UTC)
9	U1	-	min	min	Minute of hour, range 059 (UTC)
10	U1	-	sec	S	Seconds of minute, range 060 (UTC)
11	X1	-	valid	-	Validity flags (see graphic below)
12	U4	-	tAcc	ns	Time accuracy estimate (UTC)
16	I4	-	nano	ns	Fraction of second, range -1e9 1e9 (UTC
20	U1	-	fixType	-	GNSSfix Type: 0: no fix 1: dead reckoning only 2: 2D-fix
					3: 3D-fix 4: GNSS + dead reckoning combined 5: time only fix
21	X1	-	Flags	-	Fix status flags
22	X1	-	flags2		Additional flags
23	U1	-	numSV	-	Number of satellites used in Nav Solution
24	I4	1e-7	lon	deg	Longitude
28	I4	1e-7	lat	deg	Latitude
32	I4	-	height	mm	Height above ellipsoid
36	I4	-	oeLh	mm	Height above mean sea level



40	U4	1	Нс	mm	Horizontal accuracy estimate
44	U4	1	vv	mm	Vertical accuracy estimate
48	I 4	-	Nv	mm/s	NED north velocity
52	I4	-	ewE	mm/s	NED east velocity
56	I4	1	D	mm/s	NED down velocity
60	I4	-	gSpeed	mm/s	Ground Speed (2-D)
64	I4	1e-5	headMot	deg	Heading of motion (2-D)
68	U4	1	sAcc	mm/s	Speed accuracy estimate
72	U4	1e-5	headAcc	deg	Heading accuracy estimate (both motion and vehicle)
76	U2	0.01	pDOP	-	Position DOP
78	U1[6]	-	reserved1	-	Reserved
84	I4	1e-5	headVeh	deg	Heading of vehicle (2-D)
88	I2	1e-2	magDec	deg	Magnetic declination
90	U2	1e-2	magAcc	deg	Magnetic declination accuracy

6. UBX-NAV-VELNED (0x01 0x12)

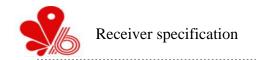
	Message		NAV-VELNED								
D	escription		Velocity Solution in NED								
		Head	Header Cla		S]	ID	Length (Bytes)	Payload	Checksum	
Mess	Message Structure		0xB5 0	0x62 0x01			0x	12	36	see below	CK_A CK_B
					Pa	yloa	ıd Co	ntent	s:		
Byte	Number	S	caling	ealing Name		me Unit		Description			
Offset	Format	D	canng	anng Name		Unit		Description			
0	U4		-	iT	OW	n	ns	G	SPS time of	week of the na	vigation epoch.



4 NvNorth velocity component **I**4 cm/s 8 East velocity component **I**4 ewE cm/s D 12 **I**4 Down velocity component cm/s speed 16 U4 cm/s Speed (3-D) Ground speed (2-D) 20 U4 gSpeed cm/s **I**4 1e-5 heading deg Heading of motion 2-D 24 28 U4 sAcccm/sSpeed accuracy Estimate 32 U4 сC deg 1e-5 Course/Heading accuracy estimate

7. UBX-NAV-SAT (0x01 0x35)

Mes	sage			NAV-SAT					
Descr	ription			Satellite Information					
		Header	Class	ID	Length (Bytes)	Payload	Checksum		
Message	Structure	0xB5 0x62	0x01	0x35	8+12*nu mxD	see below	CK_A CK_B		
			Payload	d Content	s:				
Byte Offset	Number Format	Scaling	Name	Unit	Description				
0	U4	-	iTOW	ms	GPS time of week of the navigation epoch.				
4	U1	-	version	-	Messag	e version (1 fo	r this version)		
5	U1	-	numSvs	-		Number of sat	ellites		
6	U2	-	reserved1	-		Reserved	I		
8 + 12*N	U1	-	gnssId	_	GNSS identifier for assignment		assignment		
9 + 12*N	U1	-	svId	-	Satellite identifier for assignment				
10+ 12*N	U1	-	cno	dBHz	Carrier to	o noise ratio (s	ignal strength)		

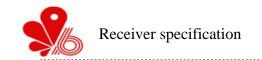


Elevation (range: +/-90), unknown if deg 11+ 12*N **I**1 elev out of range Azimuth(range 0-360), unknown if deg I2 12+ 12*N azim elevation is out of range 14+ 12*N I2 0.1 prRes Pseudo range residual m 16+ 12*N X4 flags Bitmask

8.UBX-NAV-TIMEUTC (0x01 0x21)

M	essage			NAV	/-TIMEUTC		
Des	cription			UTC	Time Solution	1	
Message Structure		Header	Class	ID	Length (Bytes)	Payload	Checksum
Message	Structure	0xB5 0x62	2 0x01	0x21	20 see below		CK_A CK_B
			Payl	oad Contents	s:		
Byte Off set	Number Fo rmat	Scaling	Name	Unit	Description		
0	U4	-	iTOW	ms	GPS time of week of the navigation epoch.		
4	U4	-	tAcc	ns	Time accuracy estimate (UTC)		
8	I4	-	nano	ns	Fraction of s	second, range -16	e9 1e9 (UTC)
12	U2	-	year	у	Yea	ar, range 199920	99 (UTC)
14	U1	-	month	month	M	Ionth, range 112	2 (UTC)
15	U1	-	day	d	Day	of month, range 1	31 (UTC)
16	U1	-	hour	h	Hour of day, range 023 (UTC)		.23 (UTC)
17	U1	-	min	min	Minute of hour, range 059 (UTC)		059 (UTC)
18	U1	-	sec	S	Seconds of minute, range 060 (UTC)		
19	X1	-	valid			Validity Flag	gs

^{9.} UBX -MON-HW (0x0A 0x09)



N	lessage			ľ	MON-HW		
De	scription			Har	dware Statu	S	
Massa	C44	Header	Class	ID	Length (Bytes)	Payload	Checksum
Messa	ge Structure	0xB5 0x6	0x0A	0x09	60	see below	CK_A CK_B
			Payload	l Contents	s:		
Byte Of fset	Number Format	Scaling	Name	Unit		Description	on
0	X4	-	pinSel	-	Mask o	of Pins Set as P	eripheral/PIO
4	X4	-	pinBank	-	Mas	k of Pins Set as	Bank A/B
8	X4	-	pinDir	-	Mask	of Pins Set as l	Input/Output
12	X4	-	pinVal	-	Masi	k of Pins Value	Low/High
16	U2	-	noisePerMS	-	Noise Level as measured by the GPS Core		l by the GPS
18	U2	-	agc	-	AGC Monitor (counts SIGHI xo SIGLO, range 0 to 8191)		
20	U1	-	aStatus	-	Machine (the Antenna Su 0=INIT, 1=DC X, 3=SHORT,4:	NTKNOW, 2=
21	U1	-	aPower	-		PowerStatus of 1=ON, 2=DON	
22	X1	-	flags	-		Flags	
23	U1	-	reserved1	-		Reserved	I
24	X4	-	usedMask	-		of Pins that are Virtual Pin Ma	•
28	U1[17]	-	VP	-	Array of	Pin Mappings for each of th 17 Physical Pins	
45	U1	-	jamInd	-		CW Jamming indicator, scaled (0 = : CW jamming, 255 = strong CW jamming)	
46	U1[2]		reserved2	-		Reserved	1

48	X4	-	pinIrq	-	Mask of Pins Value using the PIO Irq
52	X4	-	pullH	-	Mask of Pins Value using the PIO Pull High Resistor
56	X4	-	pullL	-	Mask of Pins Value using the PIO Pull Low Resistor

10. Geomagnetic output format

N	lessage			Earth's ma	gnetism				
De	scription	Output format of geomagnetic field							
		Header	Class	Length (Bytes)	Payload	Checksum			
Messa	ge Structure	0x55 0xAA	0x01	06	see below CK_A CK_F				
			Payload Co	ontents:					
Byte Of fset	Number Format	Scaling	Name	Unit	Description				
0	I1	-	YAxis low position	mxD	Geon	nagneticYaxis			
1	I1	-	YAxis height	mxD	Geon	nagneticYaxis			
2	I1	-	XAxis low position	mxD	Geon	nagneticXaxis			
3	I1	-	XAxis height	mxD	GeomagneticXaxis				
4	I1	-	ZAxis low position	mxD	GeomagneticZaxis				
5	I1	-	ZAxis height	mxD	Geor	nagneticZaxis			



2.3 Reference Program

2.3.1 NMEA0183Protocol Verification Reference Program unsigned char Calc_GPS_Sum(const char* Buffer) { unsigned char i, j, k, sum; sum = 0;for (i = 1; i < 255; i++) //ifrom1The beginning is to flash the \$ start character { if (($Buffer[i] != '*') \&\& (Buffer[i] != 0x00)) //Determine the end character {$ sum = Buffer[i]; //GPSThe checksum algorithm isXOR } else { break; } } j = Buffer[i + 1]; //Get the two characters after the terminator k = Buffer[i + 2]; if (isalpha(j)) //Determines whether the character is an English letter. If it is an English letter, it returns a non-zero value, otherwise it returns zero. { if (isupper(j)) //If the character is an uppercase English letter, it returns a non-zero value, otherwise it returns zero { i = 0x37;//Forced to16Base } else j -= 0x57;//Forced to16Base } } else if $((j \ge 0x30) & (j \le 0x39))$ { i = 0x30;//Forced to16Base } if (isalpha(k)) //Determines whether the character is an English letter. If it is an English letter, it returns a non-zero value, otherwise it returns zero. { if (isupper(k)) //If the character is an uppercase English letter, it returns a non-zero value, otherwise it returns zero { k = 0x37; //Forced to16Base } else k -= 0x57; //Forced to16Base } }



else

```
else
         {
                  if ( ( k \ge 0x30 ) && ( k \le 0x39 ) ) {
                            k -= 0x30; //Forced to16Base
                  }
         }
         j = (j \ll 4) + k; //Force merge to16Base //
         gps_sum = j;
         if ( sum == j )
                   return Valid; //Checksum OK
         }
         else
                   return Invalid; //Checksum
         Error }
    }
2.3.2 UBXProtocol Verification Reference Program
     //fromCLASSarrivePayLoadAll data
     unsigned int CalcCheckAB(unsigned char *Bytes, unsigned char len) {
             unsigned char i,a,b;
             unsigned int result;
     result = 0;
     a = 0;
     b = 0;
         for ( i = 0; i < len; i++) {
    a+=Bytes[i];
     b+=a;
```

}
result = a<<8|b;
return result;

}

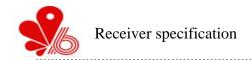
3. Contact Us

Shenzhen BroadGNSS Technology Co., Ltd. BroadGNSS Technology Co., Ltd.



Contact: Mr. Li Mobile phone: 18988798557 QQ:843570942

Address: Room 402, Shengye Space, No. 52, Tiezai Road, Bao'an District, Shenzhen.



4. Appendix 1 (Test Method)

4.1 Overview

This document primarily introduces the basic testing methods for high-precision satellite positioning receivers. The main tests include checking whether the device powers on and functions properly, verifying data output, and assessing whether the positioning accuracy is within the expected range. These tests are applicable to devices with meter-level, sub-meter-level, and decimeter-level precision.

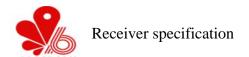
4.2 Test content

4.2.1 Test Preparation

- (1) Install the USB TO TTL serial port board driver (CH340);
- (2) Install the serial port tool on the PC, such as "sscom5.12 serial port tool";
- (3) Connect the serial port board to the locator, as shown below:

北斗GPS接收机 模块接口端	USB TO TTL 串口板			
黑线 GND	← → GNI	D		
红线 DC5V	← → DC	5V .	<>	电脑USB端口
黄线 RX	TX			
白线 TX	\longleftrightarrow RX			





- (4) Test environment with clear sky.
- (5) After completing the above steps, check the corresponding COM port driver in the Device Manager on the computer to ensure it is functioning properly, as shown in the image below:



- 4.2.2 Test steps
- (1) Open the serial port tool, such as "sscom5.12 Serial Port Tool" or HyperTerminal, and select the baud rate of 115200. The corresponding COM port number, as shown below:



If the data exceeds 1MB, check the "Receive data to file" option. Also, make sure to check "Add carriage return and line feed" before testing.

Once connected, the serial port information is as follows:

Setting parameter success Saving parameter Saving parameter success Resetting system Please wait

(2) When the serial port outputs the following information, the device shows that the positioning is successful.

\$GPRMC,033301.30,A,2236.361122,N,11351.963318,E,0.005,,280917,,,A*70 \$GPRMC,033301.40,A,2236.361121,N,11351.963311,E,0.004,,280917,,,A*7C \$GPRMC,033301.50,A,2236.361124,N,11351.963312,E,0.008,,280917,,,A*77
\$GPRMC,033301.60,A,2236.361124,N,11351.963312,E,0.004,,280917,,,A*78
\$GPRMC,033301.70,A,2236.361123,N,11351.963306,E,0.004,,280917,,,A*78
\$GPRMC,033301.80,A,2236.361123,N,11351.963306,E,0.007,,280917,,,A*77
\$GPRMC,033301.90,A,2236.361124,N,11351.963297,E,0.004,,280917,,,A*78
\$GPRMC,033302.00,A,2236.361124,N,11351.963297,E,0.006,,280917,,,A*73
\$GPRMC,033302.10,A,2236.361123,N,11351.963290,E,0.005,,280917,,,A*70
\$GPRMC,033302.20,A,2236.361123,N,11351.963290,E,0.007,,280917,,,A*70
\$GPRMC,033302.30,A,2236.361123,N,11351.963290,E,0.005,,280917,,,A*73

4.2.3 Test Example

(1) Office test: 1/4 sky static test, as shown below:

Place the RAC high-precision satellite positioning receiver on the windowsill and fix it with a bracket. Connect the laptop. Turn off the indoor signal amplifier. Repeat the test (content) steps.



Note: The measured data in this environment only verifies that the receiver can receive data normally and there is no hardware problem.

(2) Place the device on a vehicle and conduct a dynamic test while driving, as shown below:

Place the RAC high-precision satellite positioning receiver in the center of the center console under the windshield. (Note that the receiver should be placed with the front facing up) Connect the laptop. Repeat the above (test content) steps to test



(3) Placement method



Model:RAC-C1

V. Appendix 2 (Test Report)

5.1 Overview

This report describes the performance test results of the RAC high-precision satellite positioning receiver, including comparisons with RTK positioning technology under different environments. The RAC technology does not rely on any ground-based or space-based augmentation systems or inertial navigation technologies. It achieves decimeter-level dynamic positioning accuracy using only L1 or B1 frequency band signals, eliminating the need for ground-based augmentation networks in high-precision satellite positioning.

Positioning Technology	Use Geography scope	Service Fee	Data charges	External antenna	Signal blocking environment	Use Operation	High Dynamics
RAC	Any in the world Place	none	none	none	Insensitive	Easy	support
RTD	Base station broadcast scope	have	have	have	sensitive	Registration required networking	Not supported
RTK	Ground-based network scope	have	have	have	sensitive	Registration required networking	Not supported

5.2 Test platform

To evaluate the performance of the RAC high-precision satellite positioning receiver under different environments and signal strengths, we used various testing platforms, as outlined below:

Actual Signal Static Testing Platform: Testing positioning accuracy and static drift range in an open rooftop area.

Actual Signal Dynamic Road Test Platform: Conducting long-duration and long-distance testing under typical environmental conditions, including multi-lane roads, tree-lined streets, overpasses, and urban canyons.

RAC vs RTK Actual Comparison Platform: Conducting side-by-side road tests of RAC and RTK receivers under the same time and environmental conditions.

Recei

5.3 Test Results

5.3.1 Actual signal static test

In an open-sky environment, the cold start positioning time averaged 42 seconds. After static placement for five minutes, the static accuracy was within a range of less than one meter, as shown in the following image:

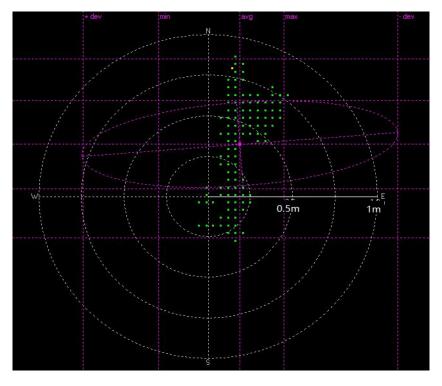


Figure 1 Static test drift trajectory

5.3.2 Actual signal dynamic test

5.3.2.1 Shenzhen multi-lane road condition test

The testing location is in Hongshuwan, Bao'an District, Shenzhen, where the lane markings are clearly visible, which can reflect the RAC lane-level accuracy performance. The receiver was placed in the center of the front windshield of the car and fixed. The vehicle was driven along different lanes. The overall route trajectory is shown below:

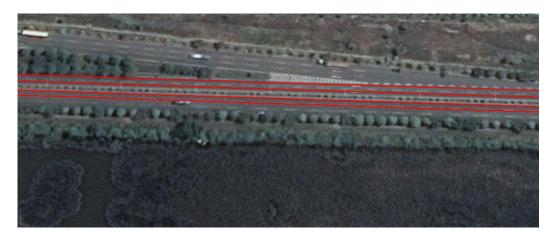


Figure 2 Shenzhen multi-lane road condition route trajectory map

5.3.2.2 Boulevard Road Condition Test

The testing location is on the 107 auxiliary road heading toward Xixiang, Shenzhen, where the road section is densely covered with trees. This test can verify the sensitivity of RAC accuracy to signal occlusion. The overall route trajectory is shown below:



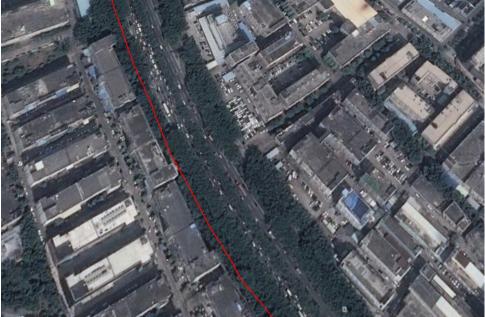


Figure 3: Boulevard road route trajectory map

5.3.2.3 Phoenix Mountain Dense Forest Test

The testing location is in the dense forest of Phoenix Mountain, Shenzhen. Four reference points were selected for positioning and data collection. The device was then moved in a circular motion around these four points. The overall average error was approximately 1.5 meters, with the error controlled within a 2-meter range. The trajectory map is shown below:



Figure 4 Phoenix Mountain Route Map

5.3.2.4 Viaduct road condition test

The testing location is near the Tiegang Reservoir in Bao'an District, Shenzhen, where the overpass length reaches 37 meters. The overall route trajectory map is shown below:





Figure 5 Viaduct road condition route trajectory map

5.3.2.5 Urban Canyon Road Condition Test

The testing location is on Gaoxin South 1st Road in Nanshan District, Shenzhen, where both sides of the test route are lined with the ZTE and Lenovo buildings. This setup allows for testing the impact of urban canyons on RAC accuracy. The overall route trajectory map is shown below:



Figure 6: Urban canyon road condition route trajectory map

5.3.2.6 Test of road conditions between high-rise buildings

Walking between two 6-story buildings, approximately 24 meters high and 240 meters long, under these extreme signal conditions, the error remains within a 3-meter range. The trajectory map is shown below:

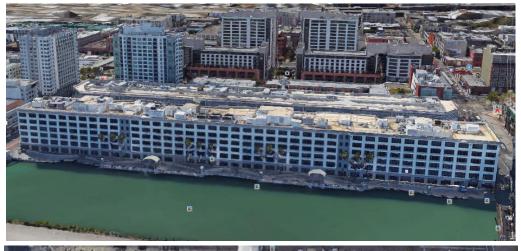
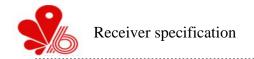






Figure 7 Route trajectory diagram of road conditions between high-rise buildings



5.3.3 RACandRTKComparison test

5.3.3.1 US Test

To verify whether RAC technology is affected by base stations or regional limitations, we selected two locations in California, USA, for testing. Figure 8 shows the road test trajectory map from Newport Center, California, and Figure 9 shows the road test trajectory map near Manhattan Beach, California.

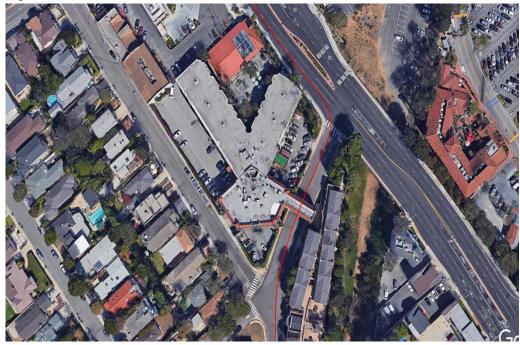


Figure 8 Road test trajectory map of Newport Center, California, USA



Figure 9 Road test trajectory near Manhattan Beach, USA

5.3.3.2 RAC, RTK, RTD dynamic comparison test

To compare the dynamic performance of RAC with RTK and RTD, tests were conducted in two locations: Beijing and Tianjin. A professional survey-grade RTK system from Trimble was used for the comparison. The specific trajectory maps are shown below:

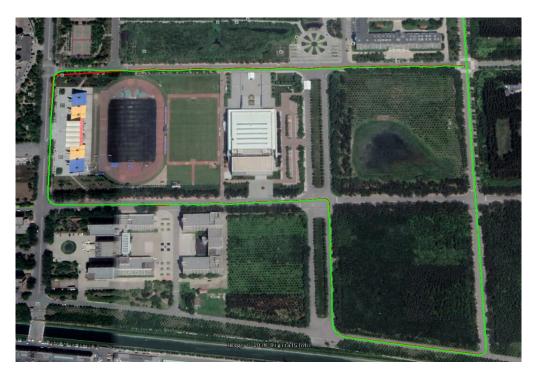


Figure 10 Road test trajectory map of Tianjin Normal University campus RAC (red) Tianbao RTK (green)

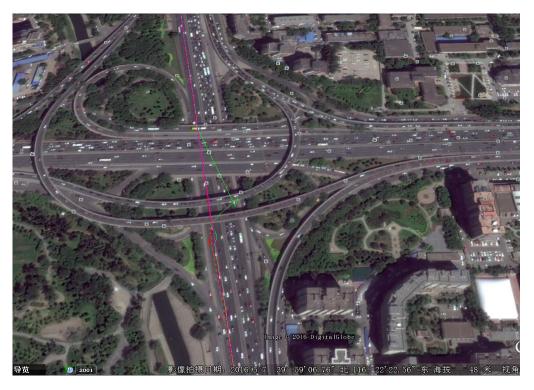


Figure 11: RAC (purple), RTD (red) and RTK (green) trajectory of a flyover in Beijing